Name:



October 5, 2022

## INSTRUCTIONS

Begin each problem in the space provided.

Write on the front side of the paper only. Work appearing on the back side of the paper will not be graded. Extra paper is available in the exam room.

If your solution does not follow a logical thought process, it will be assumed to be in error.

You must turn in your crib sheet with your exam.

#### PROBLEM No. 1 (25 points)

Problem 1 consists of 10 questions. Each question is worth 2.5 points.

(a) Machine components are often designed with multiple factors of safety.

each failure mode will have 🖊 True its non fact of safety  $\bigcirc$  False

(b) Why should factors of safety not be specified beyond one decimal place (e.g., 1.2 instead of 1.2493)

due to the uncertainty in the calculations (load, geometry, material properties all have distributions and are not search).

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- (c) Why do machine component failures occur? Select all that apply.
  - Minadequate design
  - Bad builds (fabrication issues)
  - 🖄 Insufficient maintenance
  - Difference in the second secon
- (d) Which approach to fatigue analysis is a damage tolerant approach? One that assumes cracks exist and manages the structure with periodic inspections to detect a crack before it becomes critical?
  - $\bigcirc$  Stress life
  - $\bigcirc$  Strain life
  - C Linear elastic fracture mechanics
- (e) The loading on a machine component was slowly increased until the component yielded. Yielding occurred when the loading produced a state of plane stress with  $\sigma_A = 40$  kpsi and  $\sigma_B = 30$  kpsi.

Using the MSS failure criterion, find the material's yield strength.

 $S_y = 40 \text{ Gpsi}$ 5q=40-01  $N_{y} = I = \frac{S_{y}}{\sigma_{1} - \sigma_{3}}$  $S_{y} = \sigma_{1} - \sigma_{3} = 40 - \sigma \quad \text{Kpsi}$  $O_{B=90} = \sigma_2$ 0250 Page 2 of 14

(f) The loading on a machine component was slowly increased until the component yielded. Yielding occurred when the loading produced a state of plane stress with  $\sigma_A = 40$  kpsi and  $\sigma_B = 30$  kpsi. Using the DE failure criterion, find the material's yield strength.

$$S_{y} = 36.05 \text{ kpsi}$$

$$N_{y} = 1 = \frac{s'_{y}}{\sigma'} - 3 \quad s'_{y} = \sigma'$$

$$u = \sigma'$$

$$\sigma' = \sqrt{\frac{(\sigma_{1} - \sigma_{2})^{2} + (\sigma_{2} - \sigma_{3})^{2} + (\sigma_{1} - \sigma_{3})^{2}}{2}} = \sqrt{\frac{(40 - 30)^{2} + (30)^{2} + (40)^{2}}{2}}$$

- (g) A part is subjected to cyclic loading at a frequency of 100 Hz. The part is required to be designed for infinite life. Ar does not have a life. Which material should be used?
  - AISI 1035 steel

 $\bigcirc$  A6061 aluminum allov

- $\bigcirc$  Either AISI 1035 steel or A6061 aluminum alloy can be used  $\bigcirc$  Neither AISI 1035 steel nor A6061 aluminum alloy should be used
- (h) A rotating machine component with diameter d = 40 mm has a fully corrected endurance limit of  $S_e = 340$  MPa and is subjected to torsional loading.

If the diameter was increased to d = 60 mm and all other factors remain the same, what is  $S_e$  for the component with the larger diameter?

 $S_{e} = 323 \text{ kpsi}$   $S_{e} = 323 \text{ kpsi}$   $S_{e_{j}} \text{ somm} = \frac{K_{b,60 \text{ mm}}}{K_{b,40 \text{ mm}}} \cdot S_{e_{j,40} \text{ mm}} = \frac{1.51(60)^{-0.157}}{1.24(40)^{-0.167}} \cdot 340 \text{ mmg}$   $= \frac{0.794}{0.836} \cdot 340 \text{ mmg}$ 

- (i) Fatigue failures occur due to changes in a part's material properties
  - () True False

(j) The stepped shaft shown rotates at a constant speed.

What is the stress concentration factor,  $K_t$ , at the location where load F is applied?



## PROBLEM No. 2 (25 points)

Rod OAB has length 3L and diameter d = L/8.

The rod is supported by a pin joint at O and by a roller at A.

Axial load P and transverse load 2P act at B.

The rod is made of a ductile material with yield strength  $S_y$ .

Determine the following.

- a) Solve for the reactions at O and A.
- b) Sketch and label diagrams of the internal loads on the axes provided.
- c) Identify the critical cross-section of rod OAB.
- d) Identify the critical element on the cross-section identified in part (c). You may use the attached Combined Stress Analysis Worksheet to aid your analysis.
- e) Show the state of stress on a stress element for the critical element.
- f) The factor of safety for the critical element in terms of variables P, L, and  $S_y$ . Use both the distortion energy (DE) and maximum shear stress (MSS) failure theories. If needed, axes to draw Mohr's circle are provided on the next page.



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PROBLEM No. 2 (continued)

a) 
$$\frac{\delta x}{1}$$
  
 $\frac{1}{2L}$   $\frac{1}{A_{y}}$   $\frac{1}{2P}$   
 $\Sigma F_{X}=0 \rightarrow \delta x = -P$   
 $\Sigma M_{0}=0 \rightarrow \delta y + A_{y} - \delta P \cdot 3p = 0 \rightarrow A_{y} = 3P$   
 $\Sigma F_{y}=0 \rightarrow \delta y + A_{y} - 2P = 0 \rightarrow A_{y} = -P$   
b) see axes provided  
c) critical cross-section Is just to  
right of A  
d) see worksheet. critical element IS  
 $M \to p$  (ty) of rod.  
e) see next page  
f)  $M_{y} = \frac{S_{y}}{\sigma_{1}} - \delta r DE$   
 $\sigma' = \sigma x \rightarrow N_{y} = \frac{S_{y}L^{2}}{10500P} + \int_{Total}^{Total} CM^{2}$ 

Name:

# PROBLEM No. 2 (continued)

Stress element for the critical element:





Axes to draw Mohr's circle:





### PROBLEM No. 3 (25 points)

The grooved round bar has dimensions D = 4.6 mm, d = 4 mm, and r = 0.7 mm. The bar is not rotating but the axial load cycles from 5 kN in compression to 10 kN in tension. The bar is at room temperature and 90% reliability is desired.

The bar is hot-rolled AISI 1018 steel bar with  $S_{ut} = 400$  MPa and  $S_y = 220$  MPa. Determine the following.

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- a) The fully corrected endurance limit,  $S_e$ .
- b) Draw and label the S N curve for this part. Use the axes provided.
- c) The fatigue stress concentration factor  $K_f$ .
- d) The factor of safety for infinite life using the Goodman criterion. If infinite life is not predicted, find the number of cycles until failure.
- e) Check for yielding.



## PROBLEM No. 3 (continued)



#### Figure 6–23

Fatigue strength fraction, f, of  $S_{ut}$  at 10<sup>3</sup> cycles for steels, with  $S_e = S'_e = 0.5S_{ut}$  at 10<sup>6</sup> cycles.

5=0.9 for Sut = 400 MPA



#### Figure 6–26

Notch-sensitivity charts for steels and UNS A92024-T wrought aluminum alloys subjected to reversed bending or reversed axial loads. For larger notch radii, use the values of qcorresponding to the r = 0.16-in (4-mm) ordinate. Source: Sines, George and Waisman, J. L. (eds.), Metal Fatigue, McGraw-Hill, New York, 1969.

a)  $Se = ka k_{b}k_{c}k_{d}k_{c} Se' = (0.7857)(1)(0.85)(1)(0.817).200$ = 119.8 MPa S, '= 0.5 Sut = 200 mPa  $k_a = a S_{ut}^{b} = 38.6 (400)^{-0.650} = 0.7857$ kb = 1 for axial load Kc = 0.85 "" (( kd = 1 for room temp kd = 1 Page 10 of 14 ke = 0.897 for 90% reliability

PROBLEM No. 3 (continued)

b)  $\frac{D}{T} = \frac{4.6}{4} = 1.15$  $\frac{V}{A} = \frac{0.7}{4} = 0.175$ -> /2+= 1.8  $k_{f} = 1 + q(k_{t} - 1) = 1 + 0.6(1.8 - 1) = 1.48$ q = 0.6 for r= 0.7 mm, S' = 400 mPa d)  $\frac{1}{n_{f}} = \frac{\sigma_{a}}{c_{a}} + \frac{\sigma_{m}}{S_{ut}}$  $\begin{aligned}
& \int a = K_{f} \frac{F_{a}}{\pi a^{2}/4} & \int m = K_{f} \frac{F_{m}}{\pi a^{2}/4} \\
& F_{a} = \frac{|F_{max} - F_{min}|}{2} = \frac{|2000 - (-1000)|}{2} = 1500 \text{ N}
\end{aligned}$ 

 $F_{m} = \frac{F_{max} + F_{min}}{2} = \frac{2000 + (-1000)}{2} = 500 \text{ kN}$   $T_{a} = 1.48 \cdot 4 \cdot \frac{1500 \text{ N}}{7} = 176.7 \text{ MPa}$   $T_{a} = 1.48 \cdot 4 \cdot \frac{1500 \text{ N}}{7} (0.004 \text{ m})^{2}$ 

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$$\begin{split} & \int_{m} = \frac{\sqrt{a}}{3} = 58.9 \quad mPq \\ & \frac{1}{n_{f}} = \frac{17c.7}{(19.8} + \frac{58.9}{400} \rightarrow n_{f} = 0.6 \\ & \rightarrow f_{nite} \quad life \\ & \int_{ar} = \frac{\sqrt{a}}{1-\sqrt{6m}/s_{wt}} = \frac{176.7}{1-58.9/400} = 207.2 \quad mPa \\ & N = \left(\frac{\sqrt{ar}}{n}\right)^{1/b} = \left(\frac{207.2}{1081.8}\right)^{1/-0.1573} = 32100 \\ & agdes \\ & A = \frac{(f_{s}Sut)^{2}}{s_{e}} = \frac{(0.9.400)^{2}}{119.8} = 1081.8 \quad mPa \\ & b = -\frac{1}{3}\log \frac{f_{s}Sut}{s_{e}} = -\frac{1}{3}\log \frac{0.9.400}{119.8} = -0.1593 \end{split}$$

e) 
$$n_{y} = \frac{S_{y}}{\sigma_{a} + \sigma_{m}} = \frac{220}{1767 + 58.9} = 0.9$$
  
however... when  $R_{f}$  is removed for  
Static loading  
 $n_{y} = \frac{S_{y}}{\frac{1}{1.48}(\sigma_{a} + \sigma_{m})} = 1.3 \rightarrow \text{yielding is}$   
not predicted

# PROBLEM No. 4 (25 points)

A rotating steel shaft is simply supported by bearings at A and C. The diameter of the larger section is 1.5 inches and the diameter of the smaller section is 1 inch.

The shaft is loaded with constant transverse force of P=300 lbf at point B, and a torque that alternates between  $T_{min} = -100$  lbf-in and  $T_{max} = +700$  lbf-in in section BC.

The shaft has the following material properties:  $S_{ut} = 100$  kpsi,  $S_y = 62$  kpsi,  $S_e = 30$  kpsi (fully corrected).

The fatigue stress concentration factors at the step are  $K_f = 2.1$  for bending and  $K_{fs} = 2.5$  for torsion.

Determine the following.

- a) Solve for the reactions at A and C.
- b) Sketch and label diagrams of the internal loads on the axes provided.
- c) Identify the critical cross-section of the shaft.
- d) The factor of safety for infinite life using the Goodman criterion. If infinite life is not predicted, find the number of cycles until failure.
- e) Check for yielding.



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PROBLEM No. 4 (continued)



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**PROBLEM No. 4** (continued)

 $e^{-}$   $n_{y} = \frac{5y}{\sigma_{a}^{1} + \sigma_{m}^{-}}$   $n_{y} = \frac{62}{(13.35)} = 2.5$